

EUCAS 2013, Genova, Italy

Superconducting detectors for Millimetron space mission

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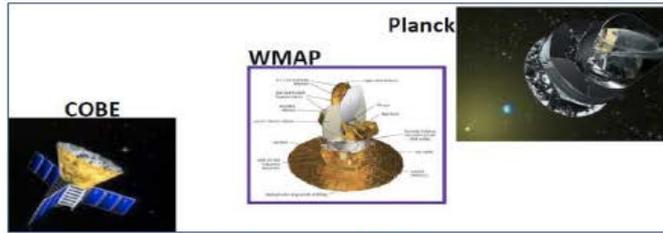
university of
 groningen





IR Space Missions overview

CMB missions



Spectral Survey



Astro-F



SPITZER



2004-2009



1983-84



1995-1998

SWAS Odin



WISE



2013



2018

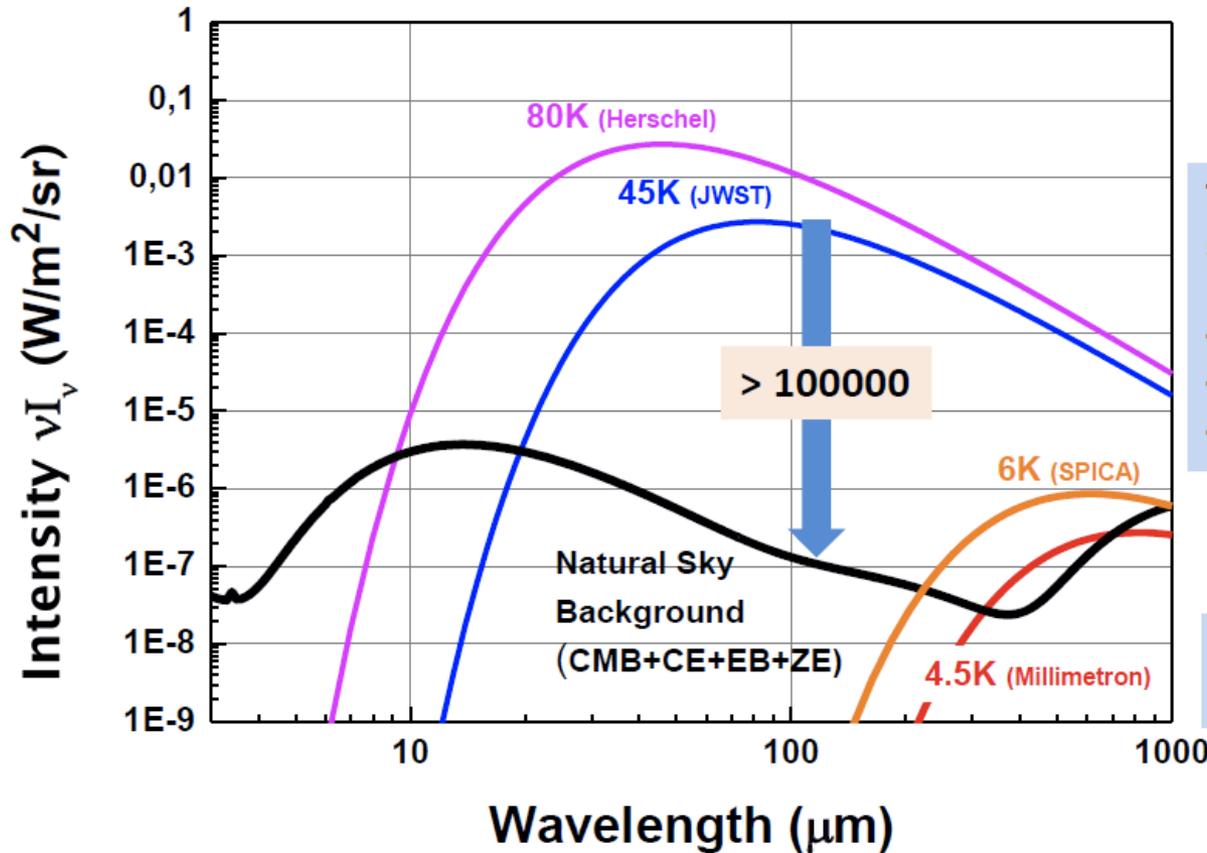


2011;SOFIA



Scientific requirement for next space missions: low background

Cold telescope



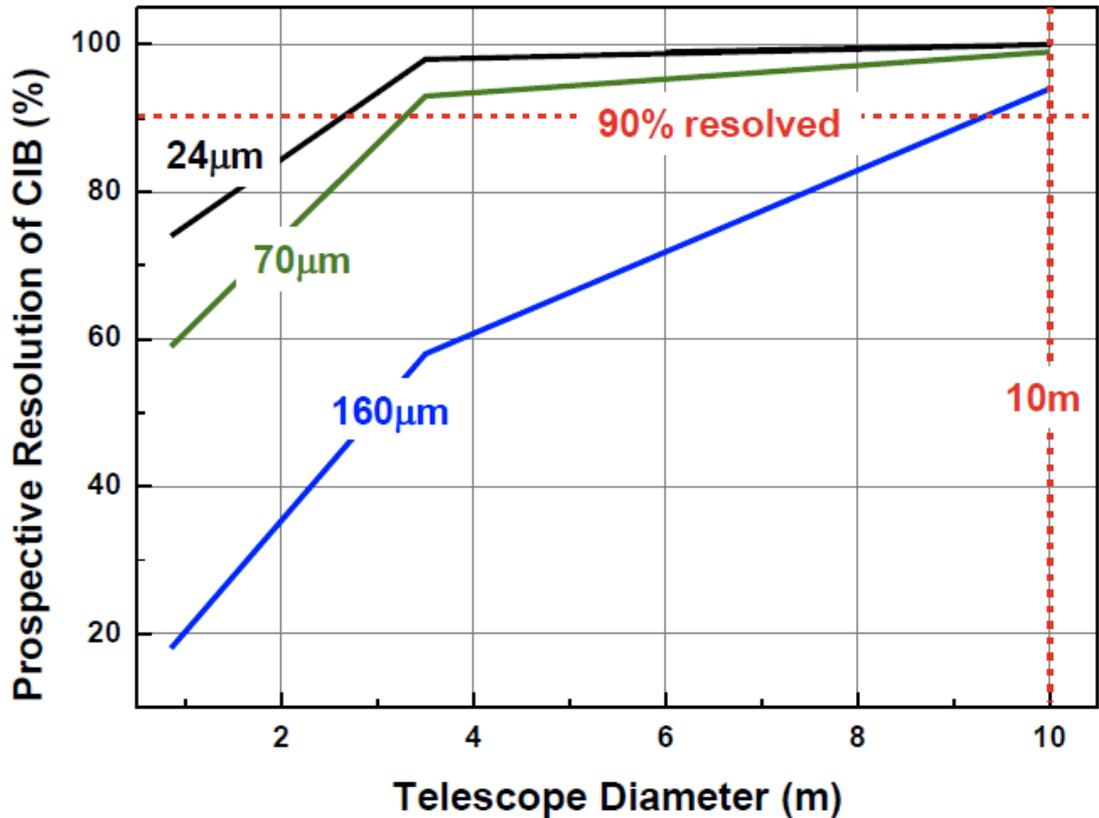
The intensity of the thermal self-emitted radiation falls below the limit provided by the natural sky background if the telescope is cooled to temperature < 10K



Background limited < 300 μ m if $T_{tel} \approx 4.5$ K



Scientific requirement for next space missions: high spatial resolution

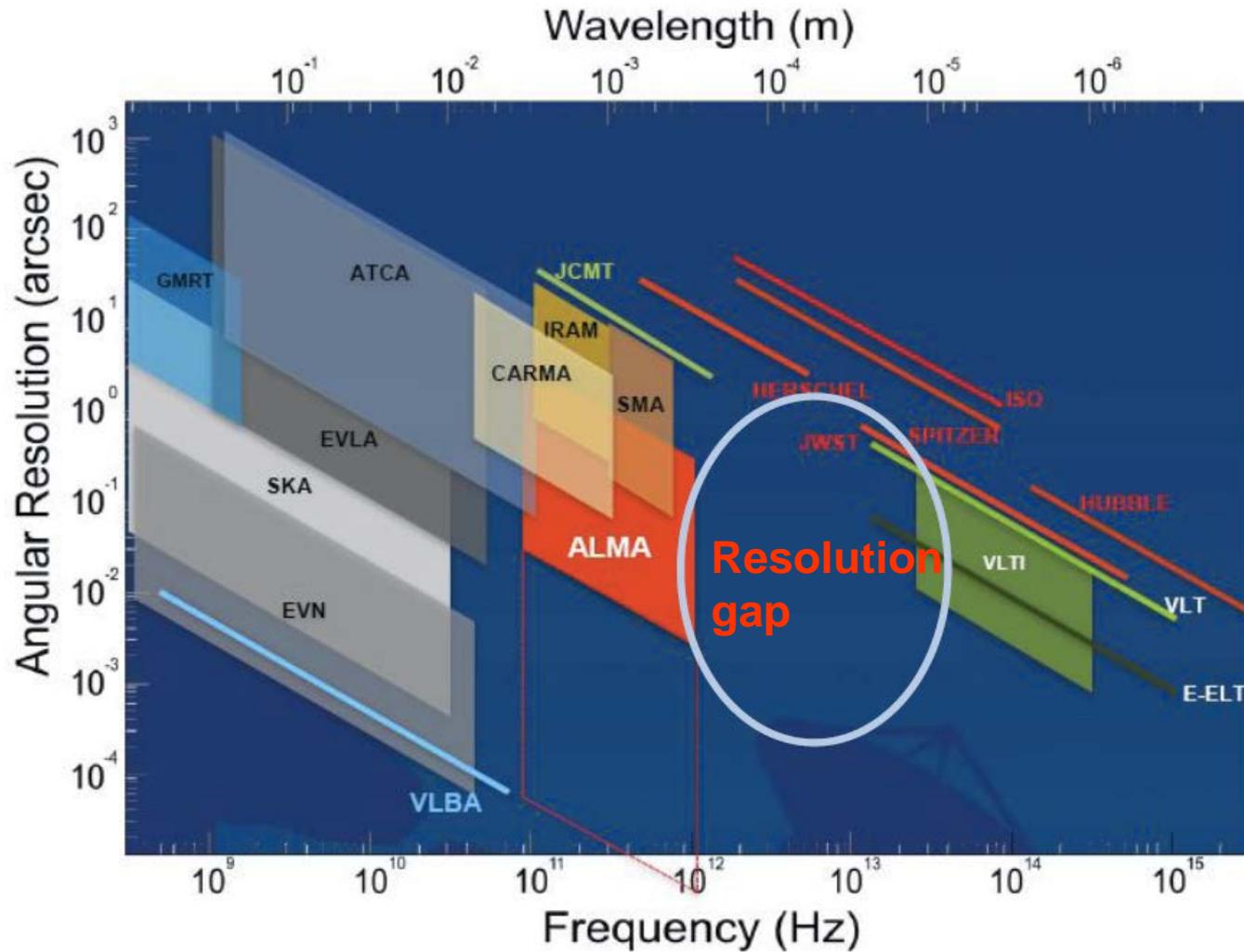


The telescope with a big aperture can resolve more than 90% CIB into individual sources

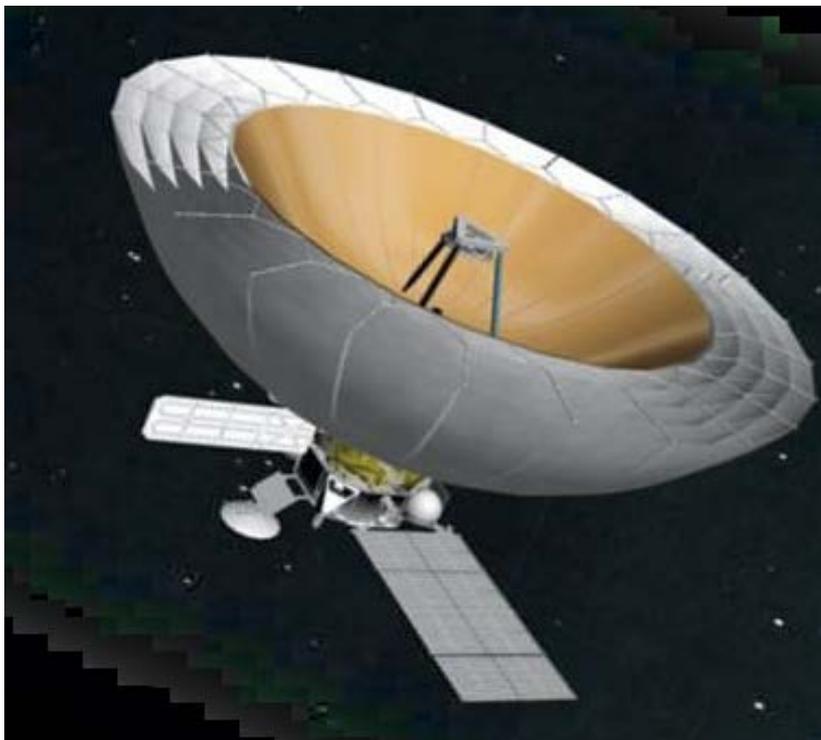
* Dole, H., Rieke, G. H., Lagache, G., et al., ApJ, 154, 93–96 (Sept. 2004)



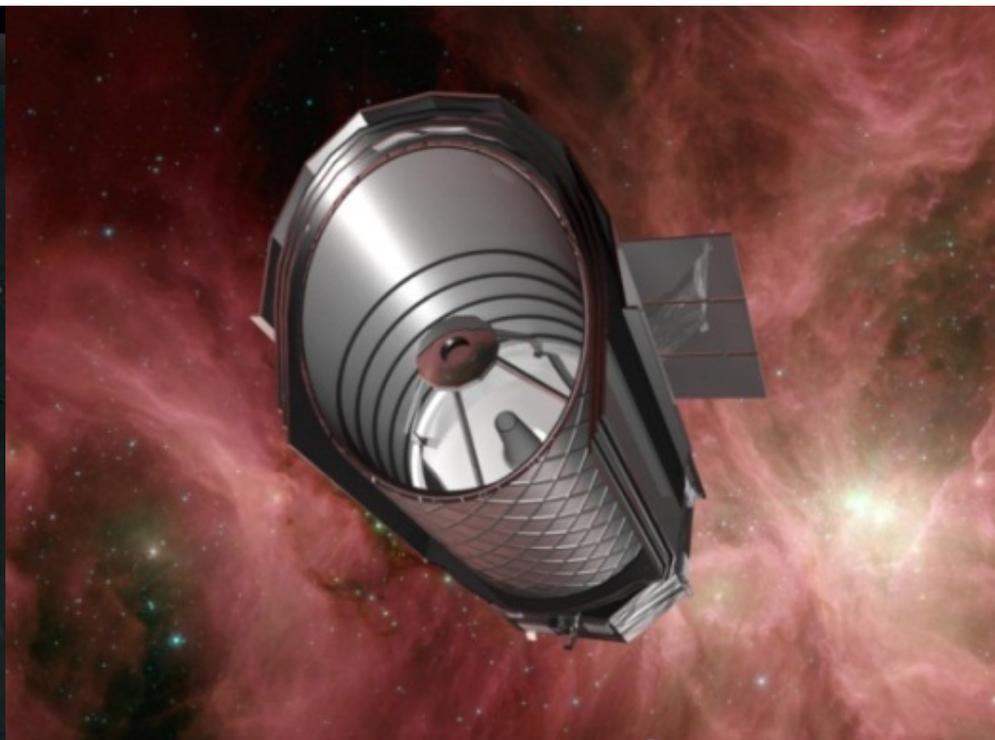
High spatial resolution



Near future space missions: both cold



Millimetron/Spectrum-M

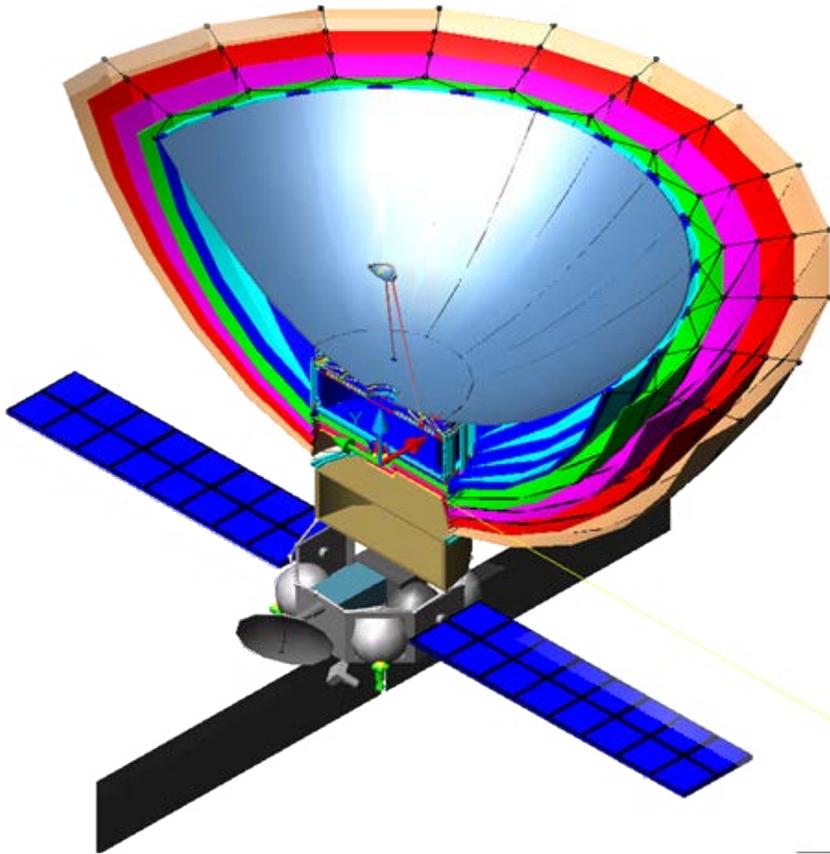


SPICA

Both require very sensitive detectors



☐ Millimetron mission overview



First 10 m class space telescope for - the FIR, submm and mm range

- deployable and adjustable
- mechanically cooled ($<10\text{K}$) with post-cryo life
- L2 orbit - for cosmology and astrophysics

Dual operation modes:

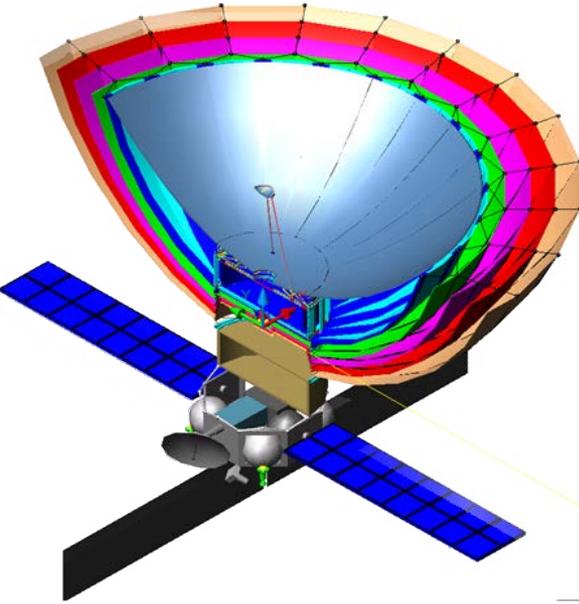
- S-VLBI up to 1 THz
- single dish with observatory-style
- largest array formats flown to-date
- state-of-the-art instrumentation with extensive space technology heritage

Launch date aimed for 2019/2020

Lifetime: 10 years; at cryo >3 years



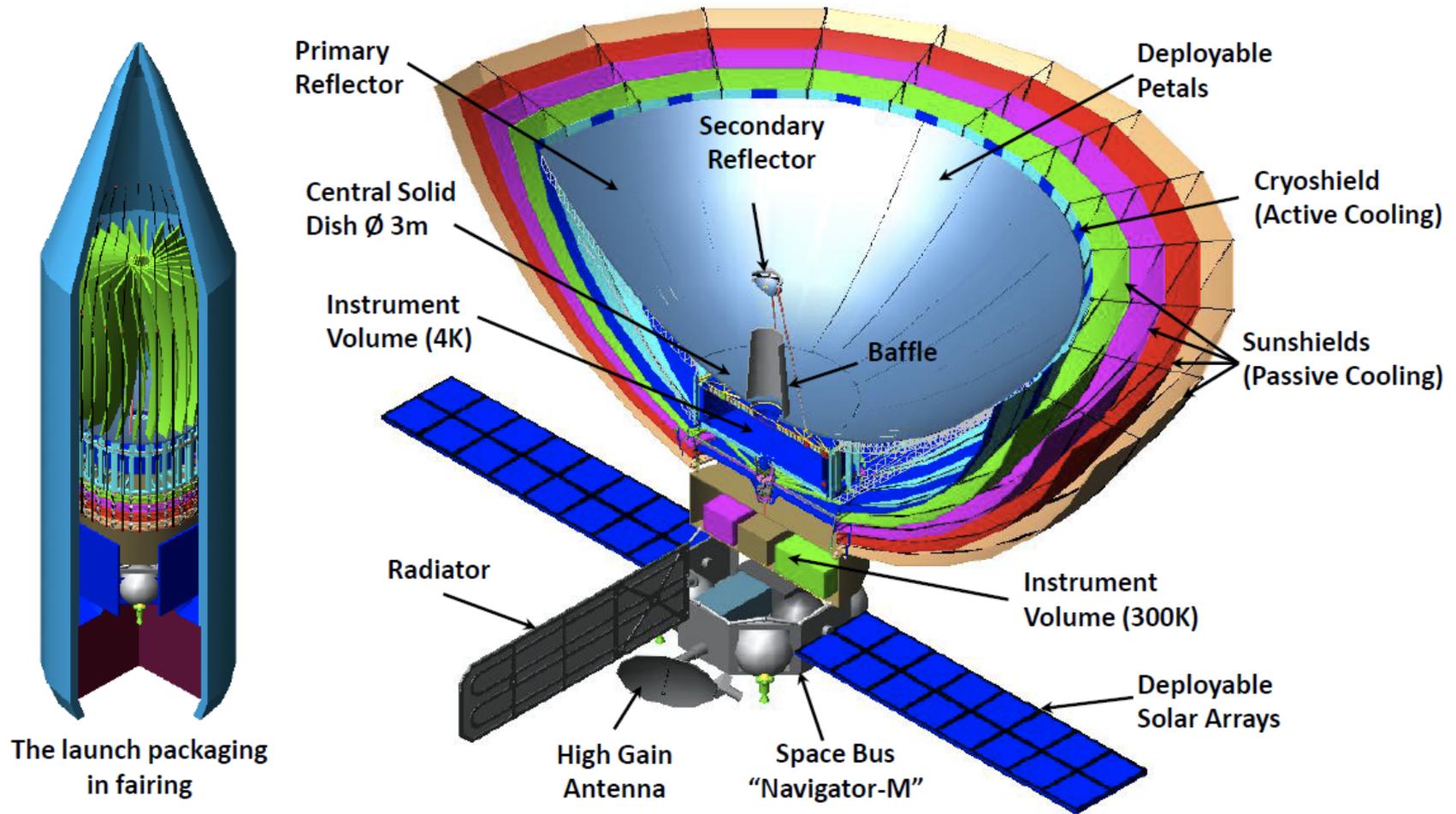
Millimetron mission overview, contd.



- Wavelength range: 50 (goal)/150 μm – 3mm/17mm(S-VLBI)
- Two modes of observation: S-VLBI element and as single-dish telescope
- Instruments: Heterodyne and Direct detection (arrays)
- Cassegrain system, Primary with aperture $\text{\O}10\text{m}$ ($\leq 10\mu\text{m}$ rms) consist of central solid dish $\text{\O}3\text{m}$ ($\leq 5\mu\text{m}$ rms) and deployable petals
- Cooling: radiation + mechanical coolers
- Telescope temperature: $< 10\text{K}$; 4.5 K goal.
- Total mass: ≤ 6600 kg
- Total electrical power : ≤ 4000 W
- Orbit: L2 for the Sun-Earth system
- Lifetime: 10 years (3 years with active cooling)



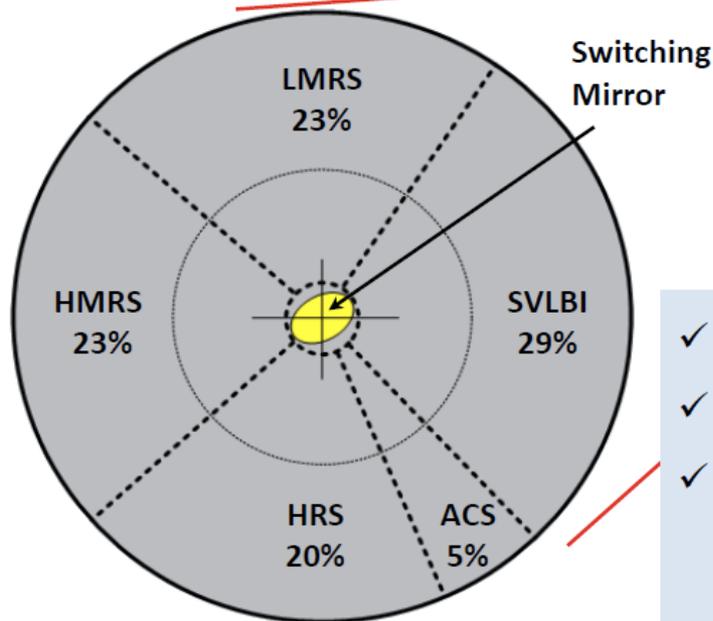
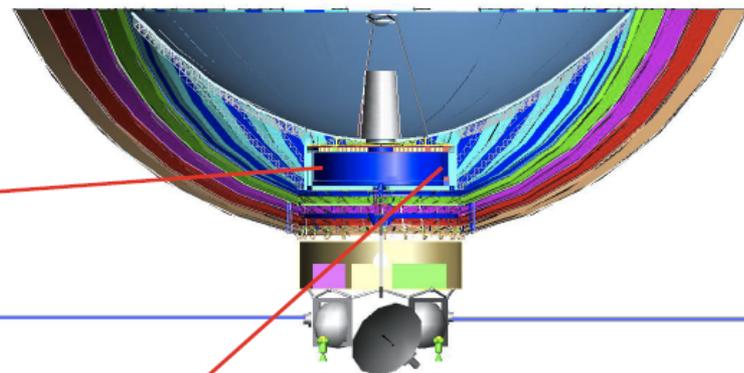
Millimetron mission concept



Resources available on Millimetron



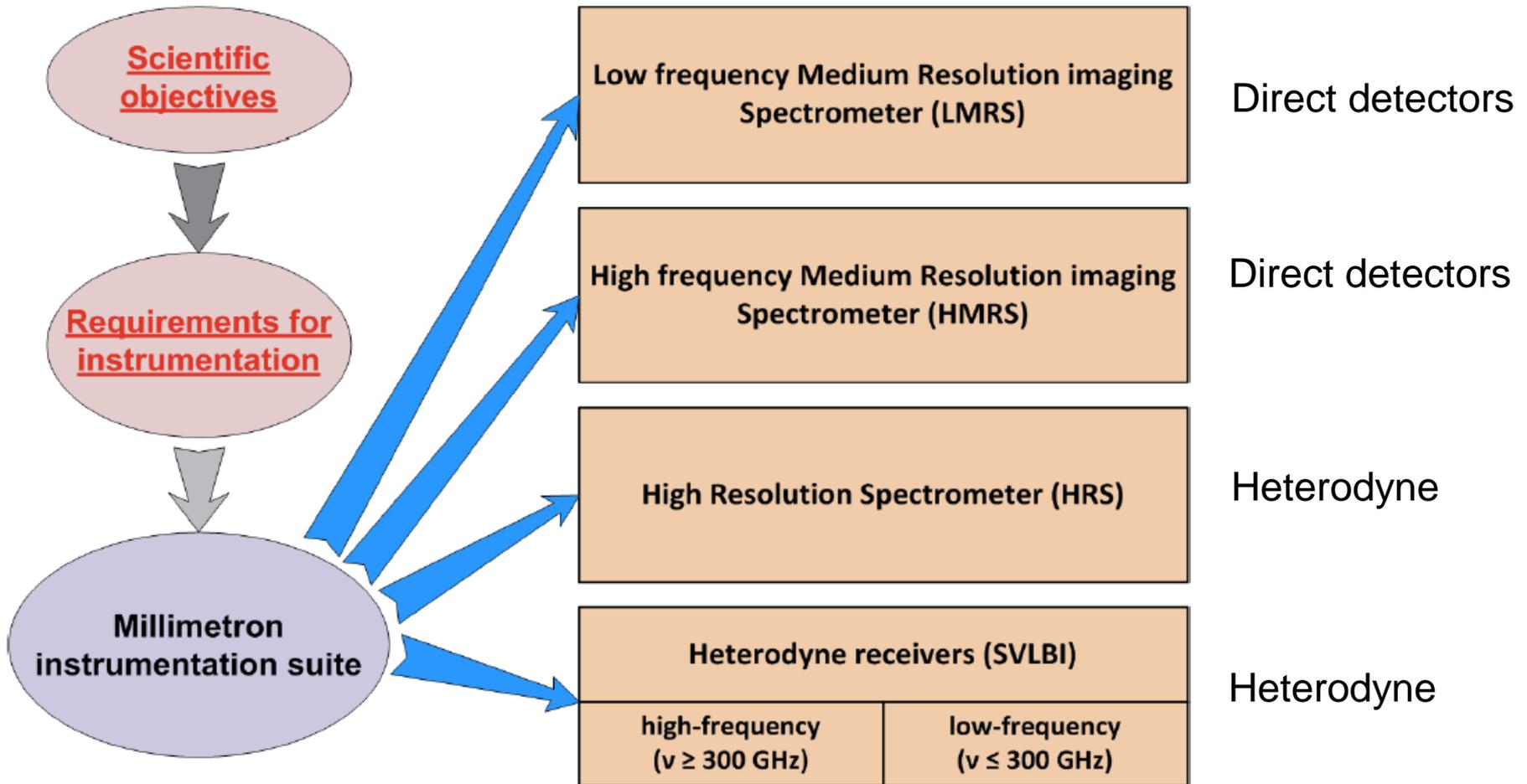
A preliminary sketch of the assembly of instruments in the cryogenic volume



- ✓ Instrument Volume: 2.5 0.8 m (diameter height)
- ✓ Cold Mass: \approx 300 kg (with margin)
- ✓ Thermal loads :
 - \leq 100 mW @ 4.5 K stage
 - \leq 15 mW @ 1.7 K stage



Millimetre Instrumentation



☐ Detector technology requirements

- High Sensitivity
- Low mass
- Low power consumption (also for readout)
- Low power dissipation both ambient and cryogenic level
- Low wire count for many detector pixels
- Vacuum compatibility
- **Low sensitivity for cosmic ray hits**

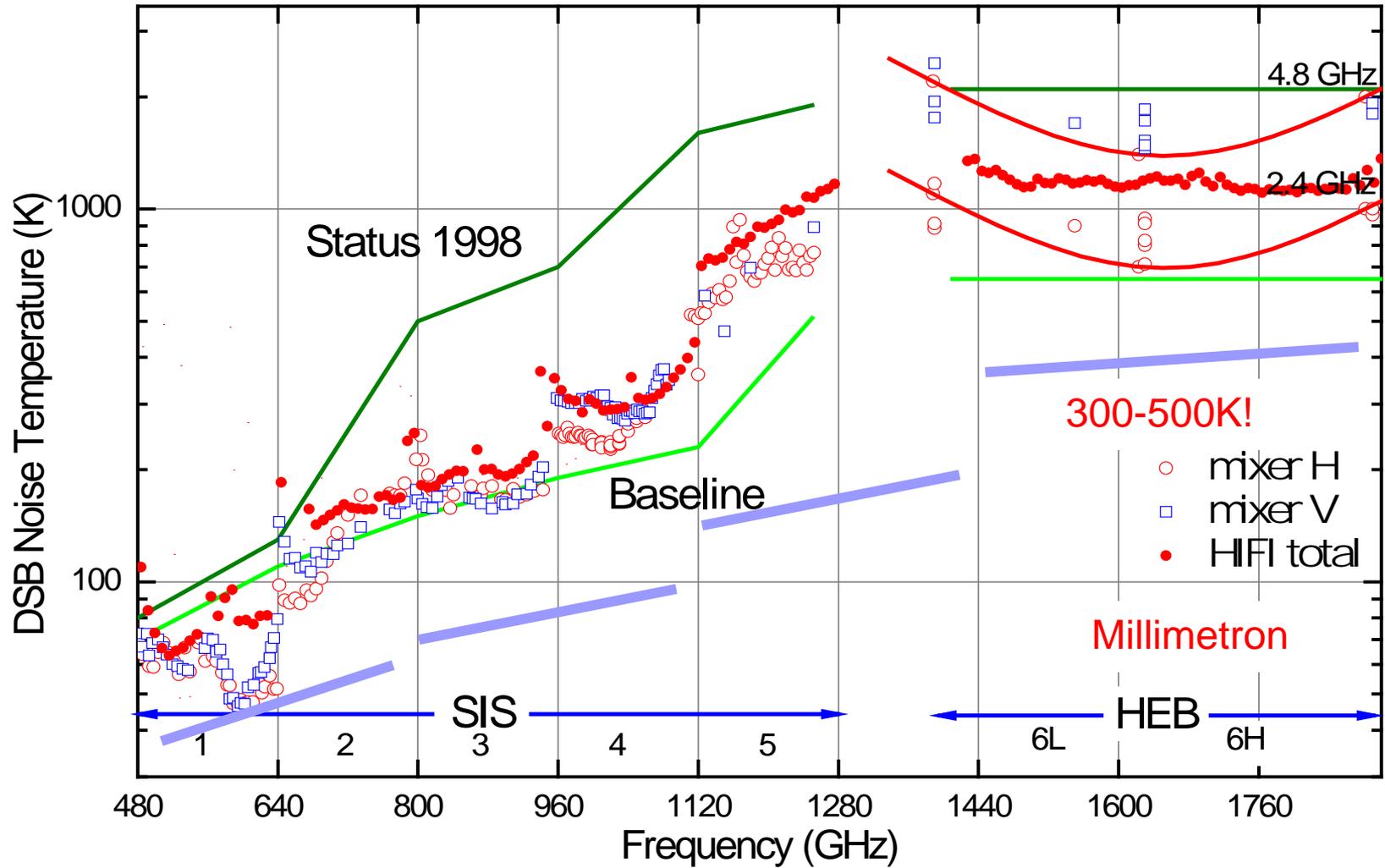
☐ Heterodyne instruments coverage

8 GHz IF bandwidth, x 16 in peak configuration

Band	Frequency (GHz/THz)	IFBW (GHz)	Polarization	Array size
1a 1b	275 – 373	4-12 (SIS)	H V	Single pixel
2a 2b	557 – 752	4-12 (SIS)	H V	Triangular (baseline) Hexagonal (goal)
3a 3b	752 – 950	4-12 (SIS)	H V	Hexagonal
4a 4b	0.95 – 1.15 1.15 – 1.40	4-12 (SIS) 1-8 (HEB)	single	Hexagonal
5a 5b	1.40 – 1.80 1.80 – 2.10	1-8 (HEB)	single	Hexagonal
6a 6b	2.45 – 3.00 4.76 – 5.36	1-8 (HEB)	single	Hexagonal

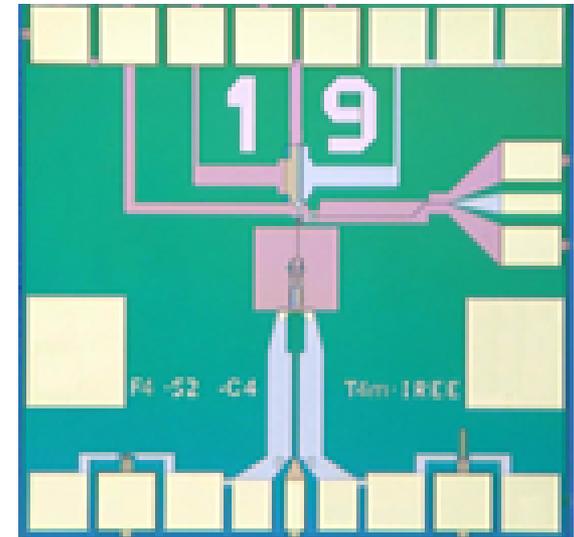
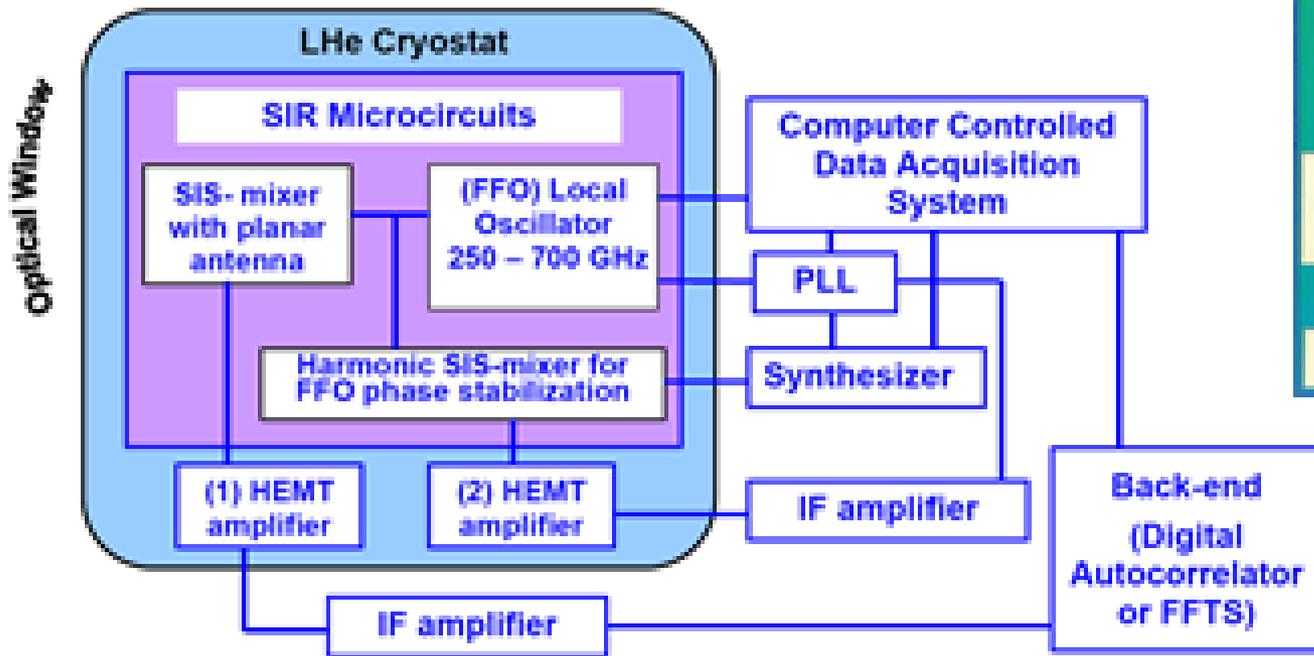


Heterodyne sensitivity





Superconducting Integrated Receiver (SIR) with phase-locked FFO



**SIR
microcircuit**



Nb–AlO_x–Nb или Nb–AlN–NbN
 Current density : $J_c = 3 - 8 \text{ kA/cm}^2$;
 Tunnel barrier thickness : $\sim 1 \text{ nm}$;
 SIS junction area: $\sim 1 \text{ }\mu\text{m}^2$

SIS technology

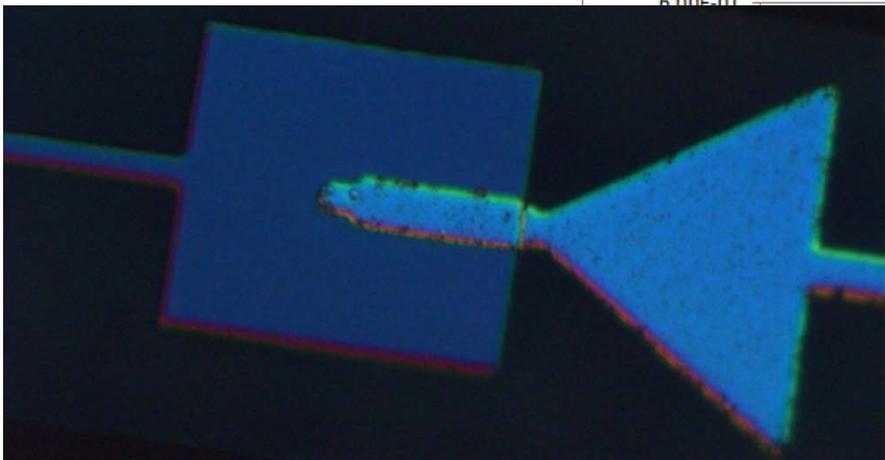
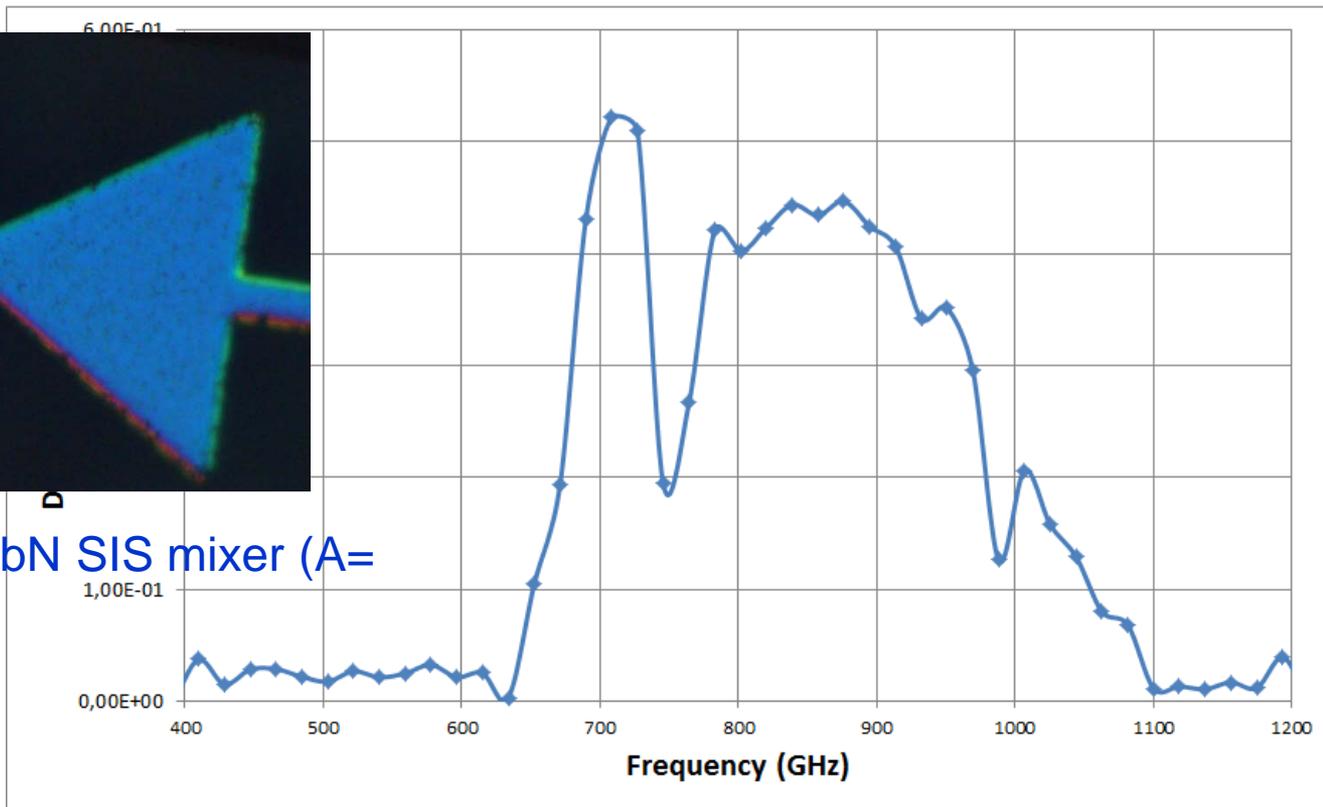


Photo of the Nb/Al-AlN_x/NbN SIS mixer (A= 0.5 μm²)

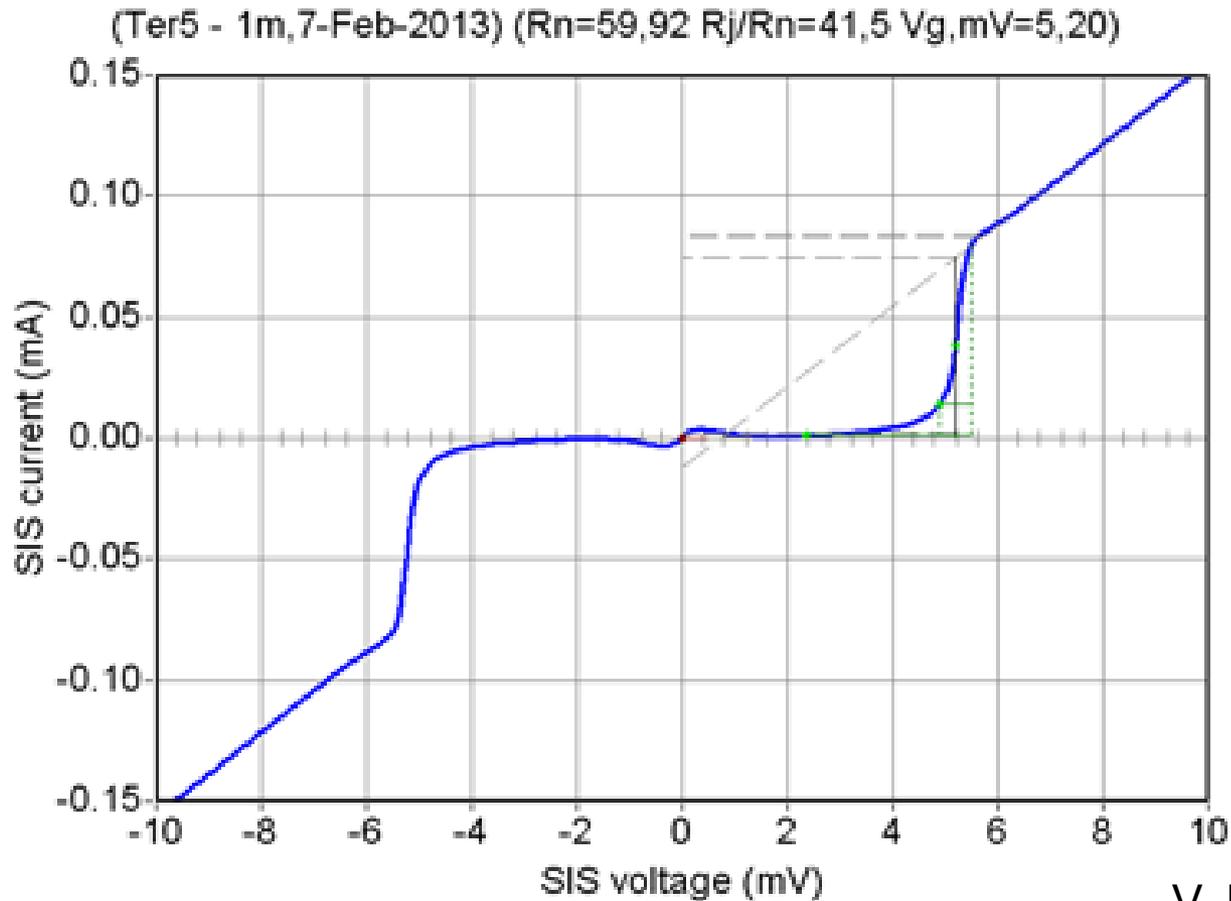


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NbN-MgO_x-NbN SIS on the MgO substrate; tunnel barrier made by plasma oxidation of the 1.5 nm Mg layer



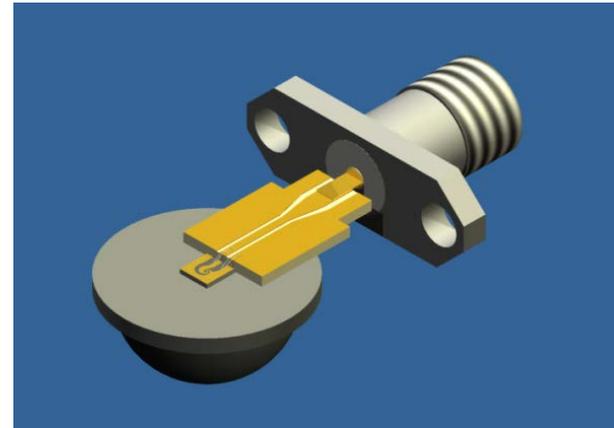
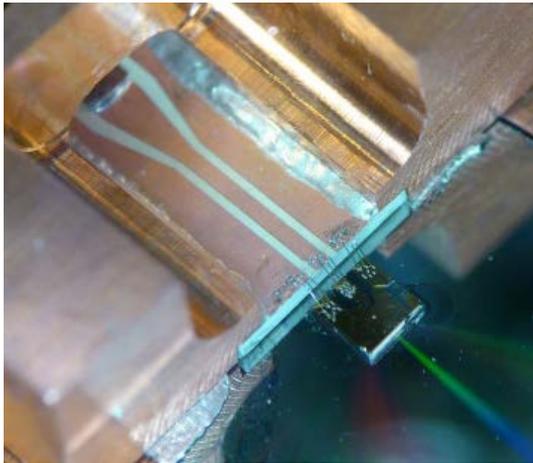
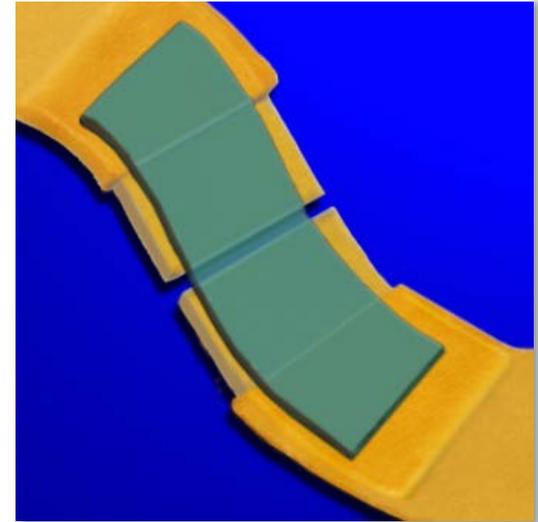
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Hot Electron Bolometer



- NbN $2 \times 0.2 \mu\text{m}^2$
- Quasi-optical,
- Tight spiral antenna
- Coated Si lens for 4.3 THz



Gao et al

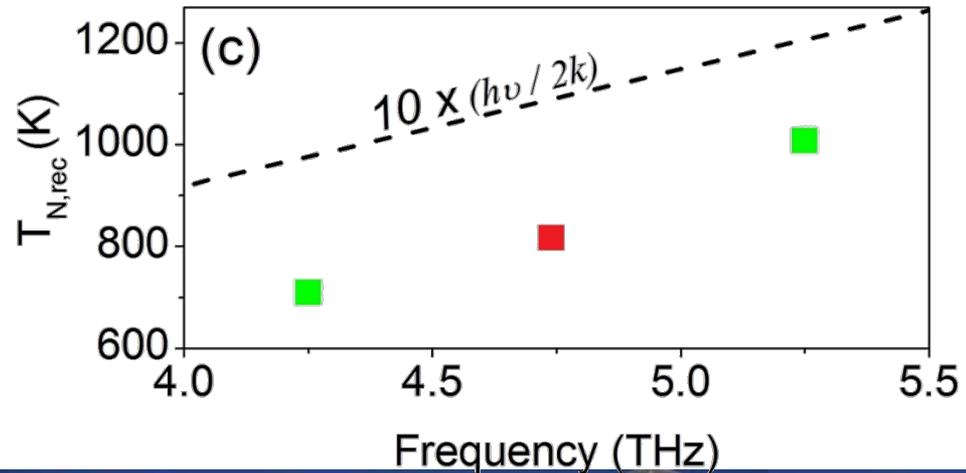
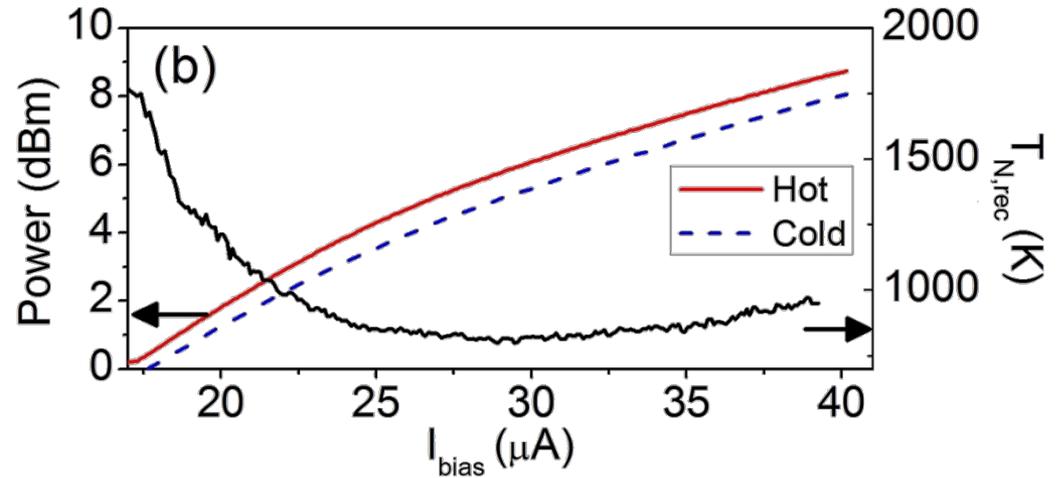




HEB-QCL at 4.7 THz

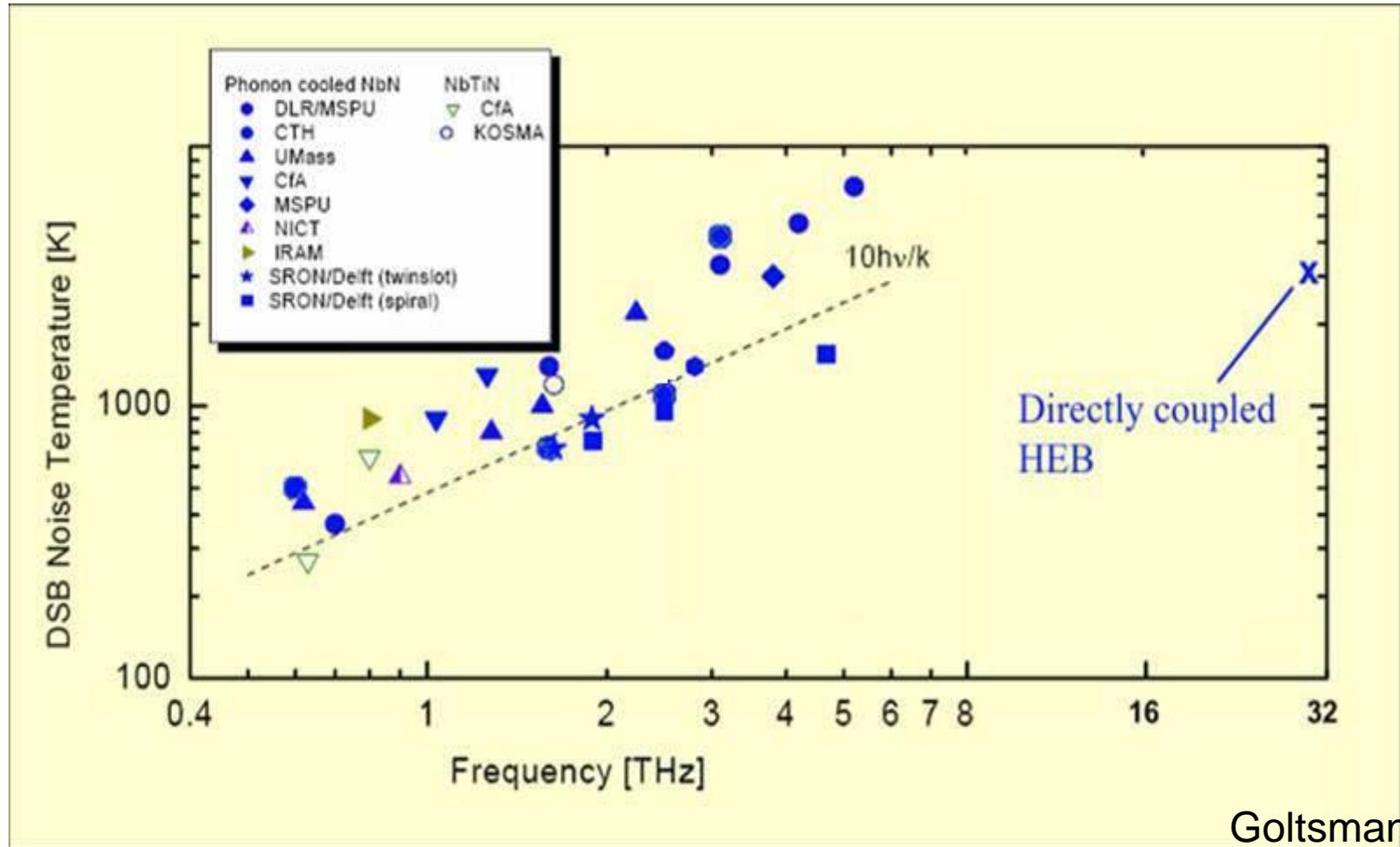
Lab setup for 4.7 THz receiver

- Measured $T_{N,rec}$, DSB of 815 K with 4.7 THz QCL (3 μm beamsplitter)
- ~ 7 times quantum noise
- Red point: QCL LO
- Green points: FIR gas laser



J. L. Kloosterman et al., Appl. Phys. Lett. 102, 011123 (2013)

Sensitivity overview

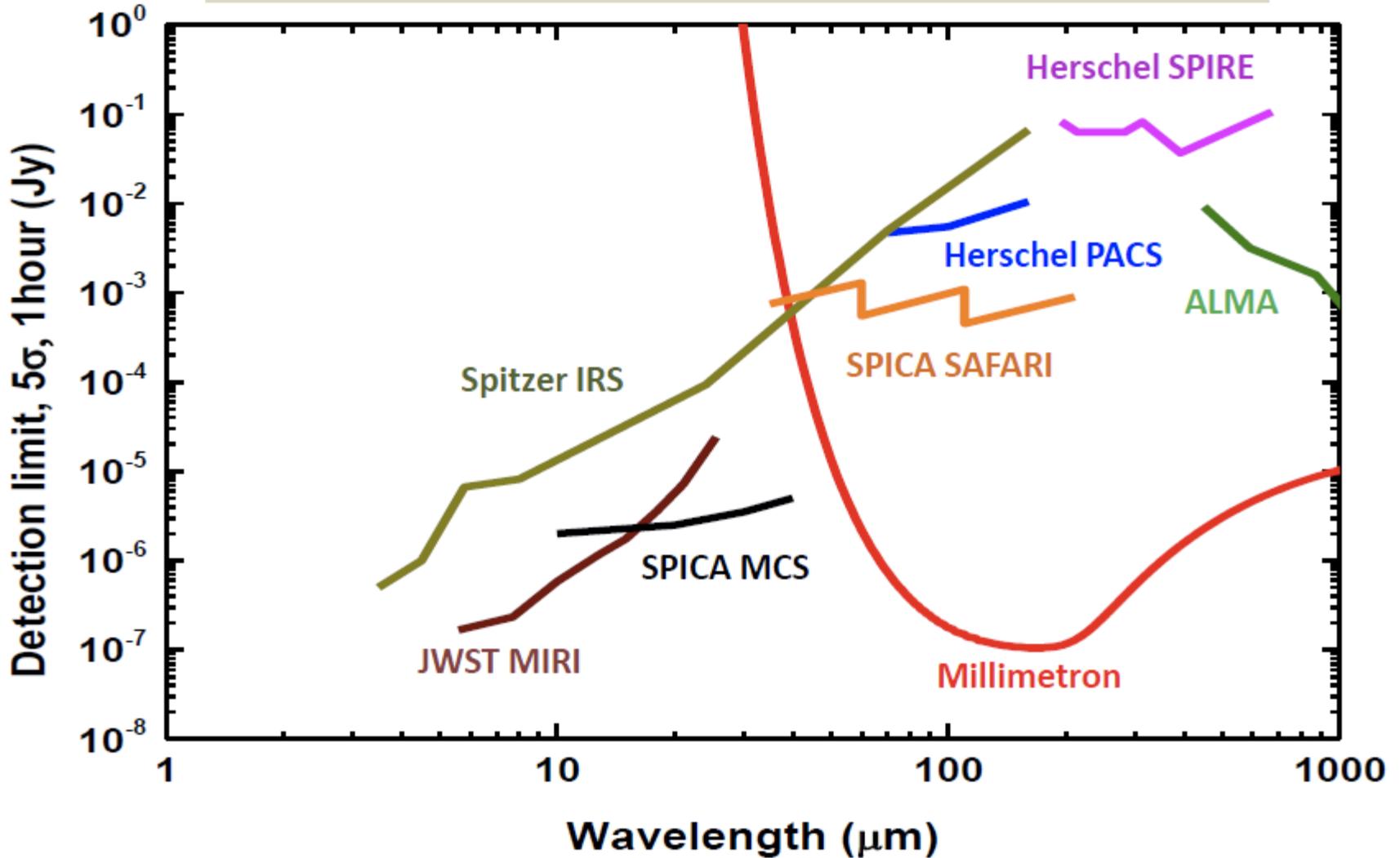




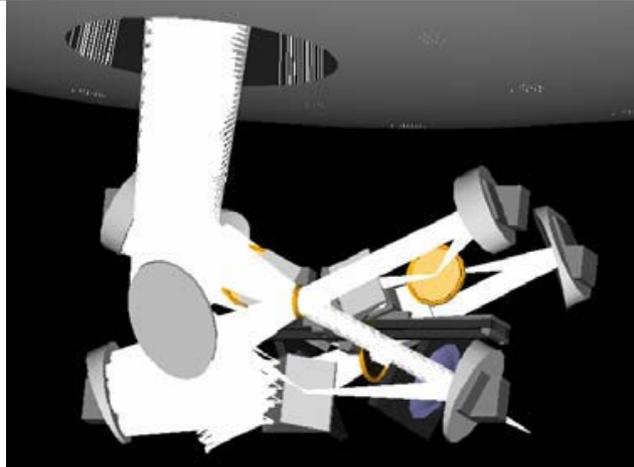
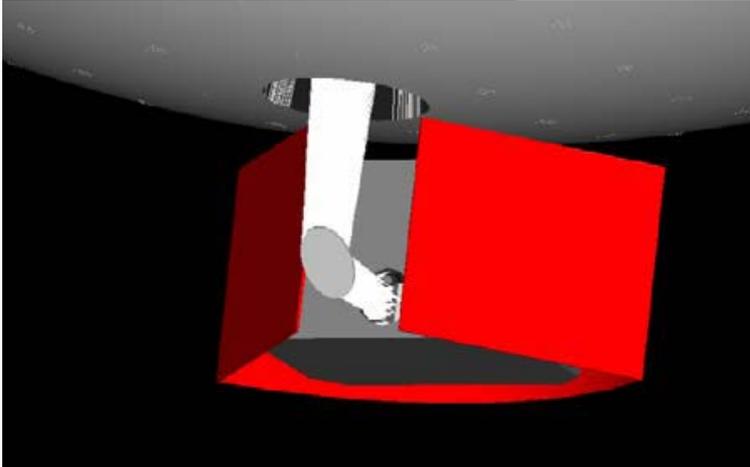
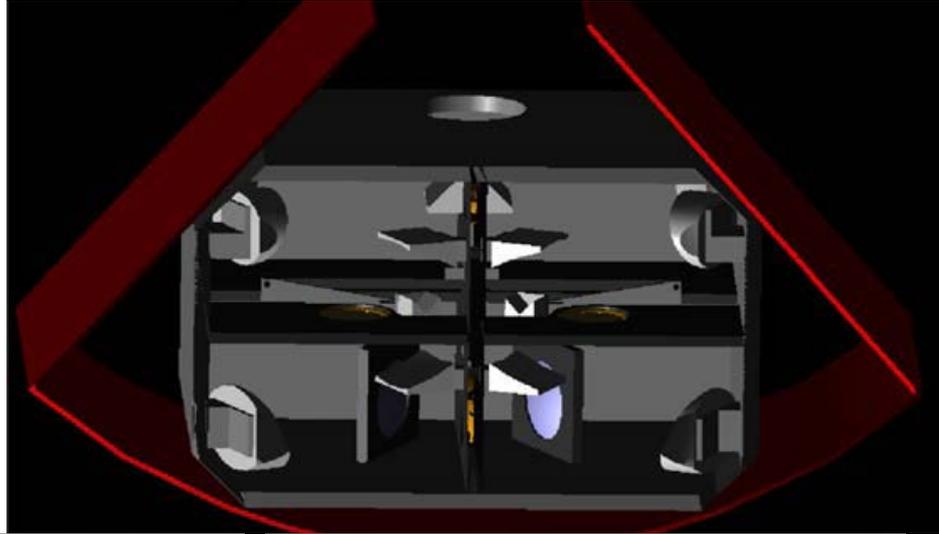
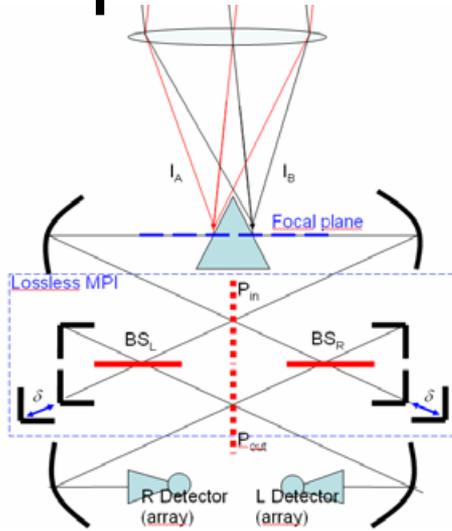
Direct detectors

- Microwave Kinetic Inductors
 - Cold Electron Bolometers CEB
 - Transition edge sensors
-
- NEP $<1\text{E}-19$ Goal $1\text{E}-21$, for grating spectrometer

Spectroscopic sensitivity, grating



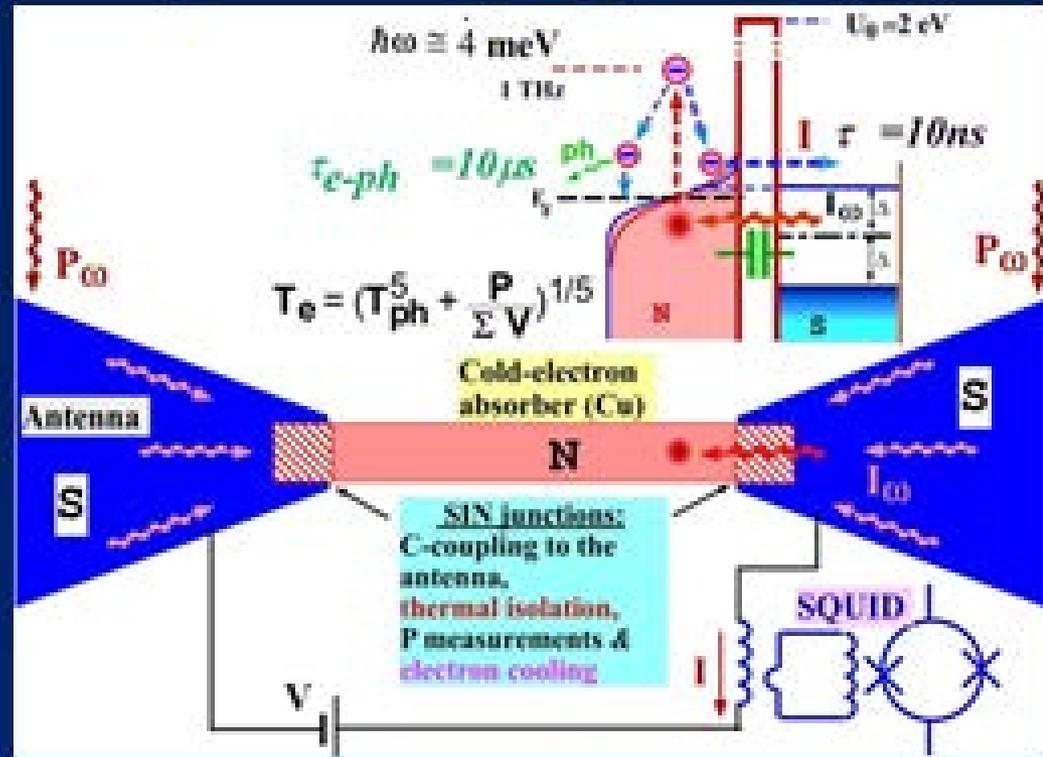
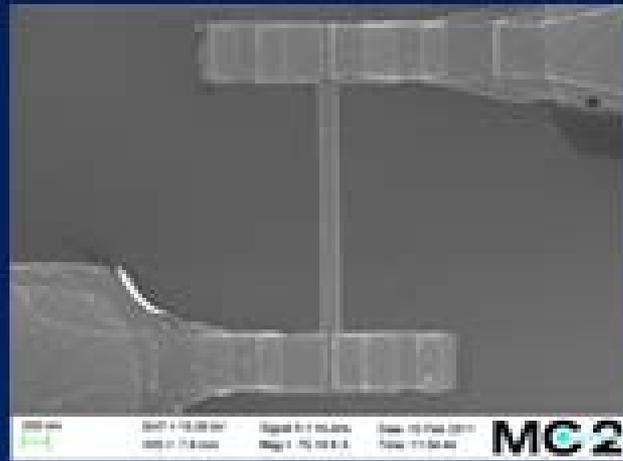
Medium resolution low frequency spectrometer layout



P. De Bernardis et al



Cold-Electron Bolometer (CEB) with Capacitive Coupling to the Antenna



Main features of the CEB:

1. High sensitivity **due to electron cooling effect:**

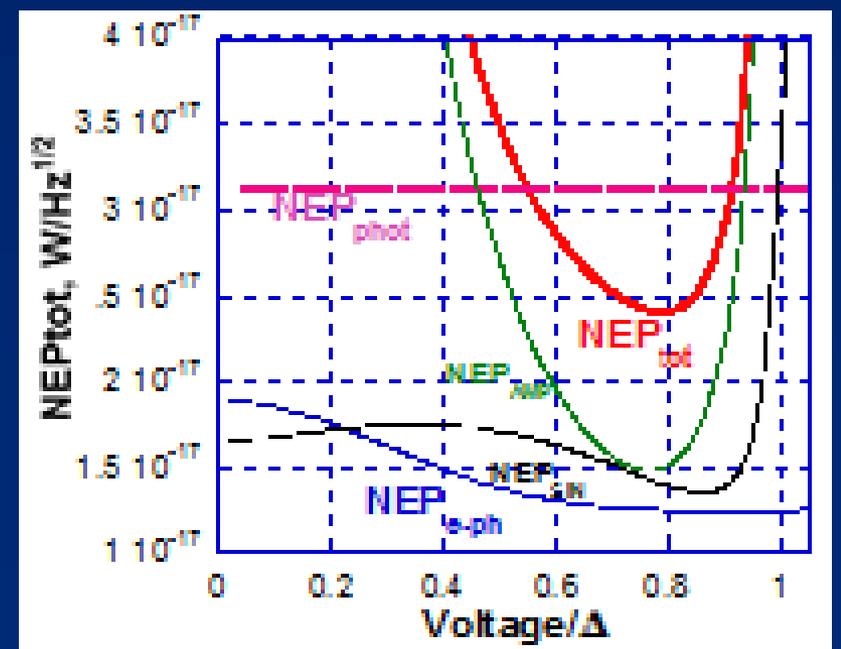
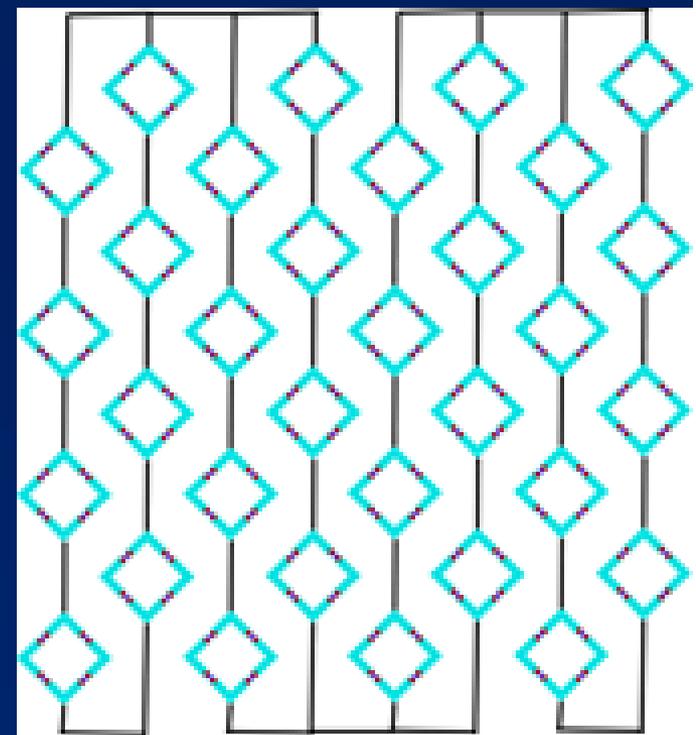
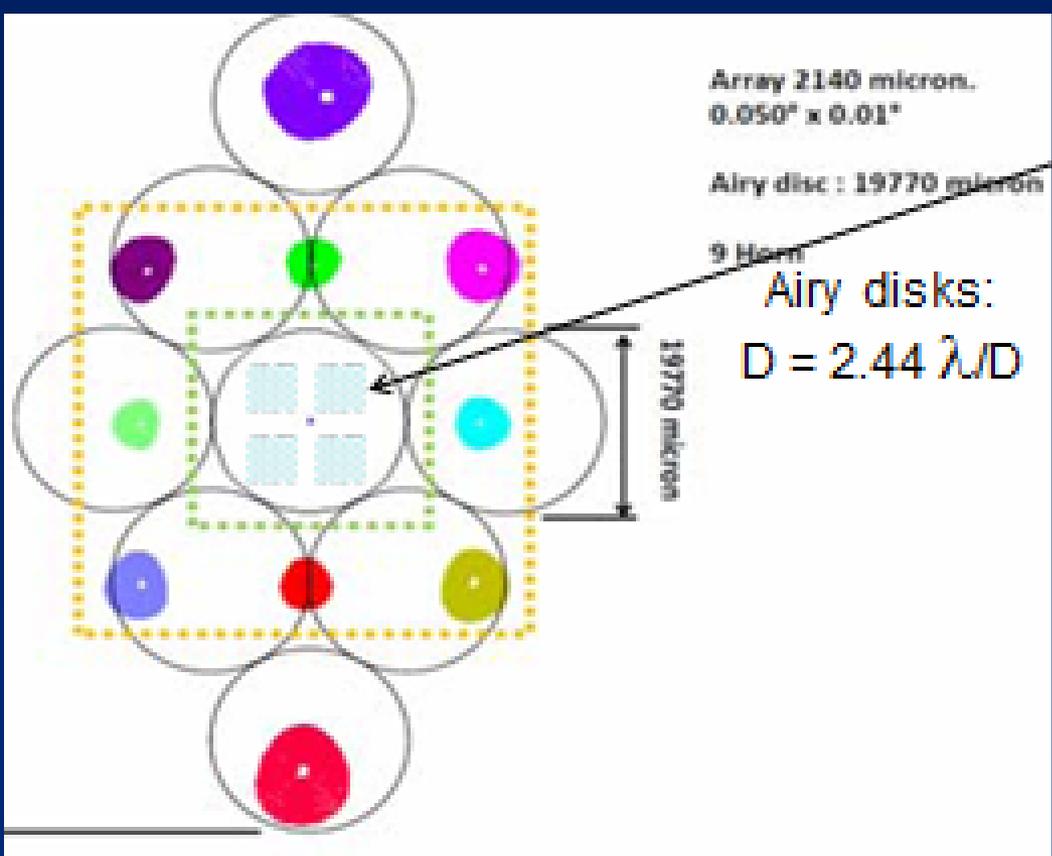
$NEP_{ceb} < NEP_{ph}$ for any optical power load (from 80 pW for LSPE to 0.02 fW for SPICA)

2. High dynamic range **due to direct electron cooling**

3. Insensitivity to Cosmic Rays (CR)

4. Very easy to fabricate in arrays on planar substrate

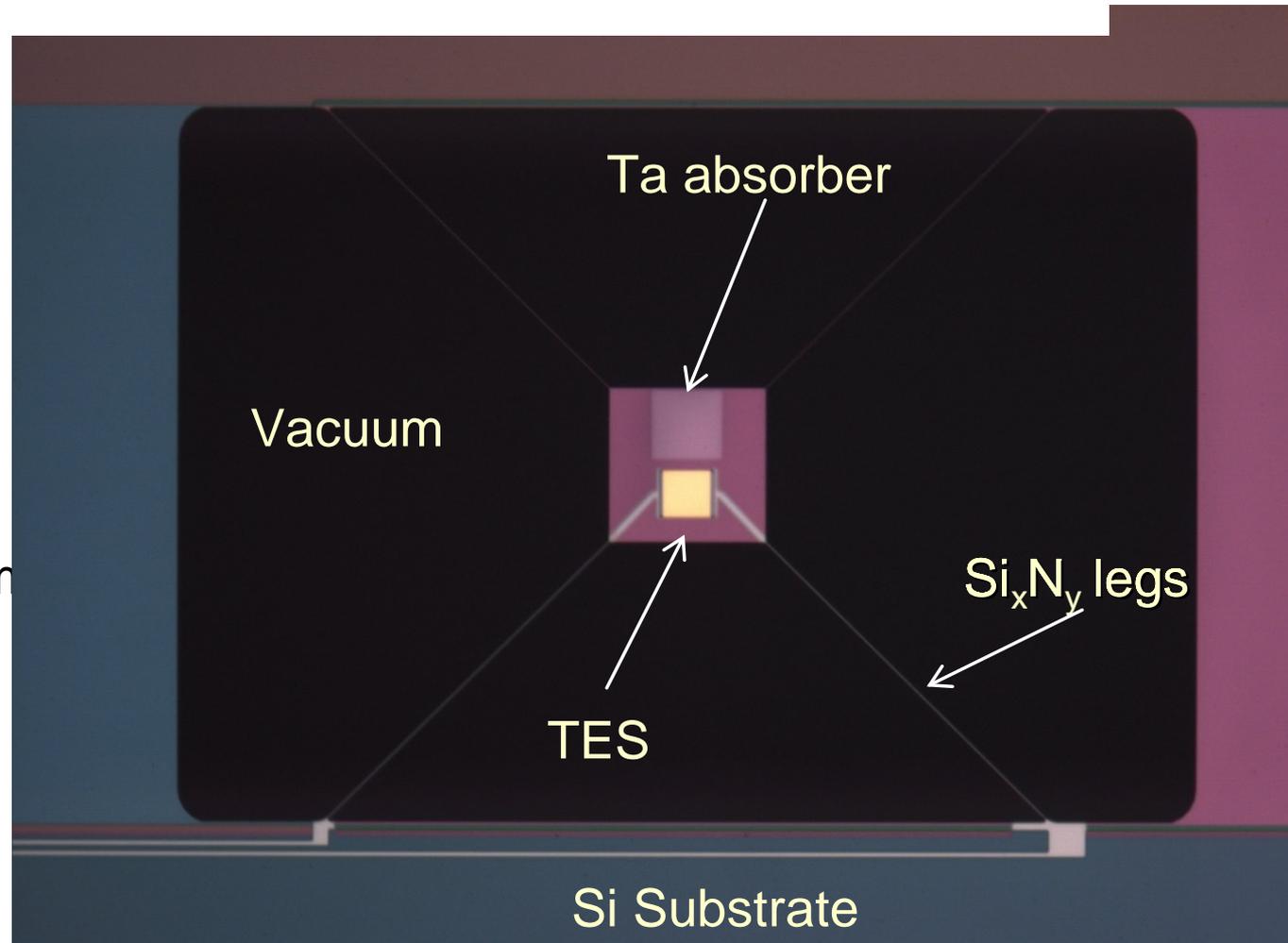
4 CEB Arrays with Dual-Polarized Antennae in each Airy disk



Transition edge sensors, SPICA SAFARI

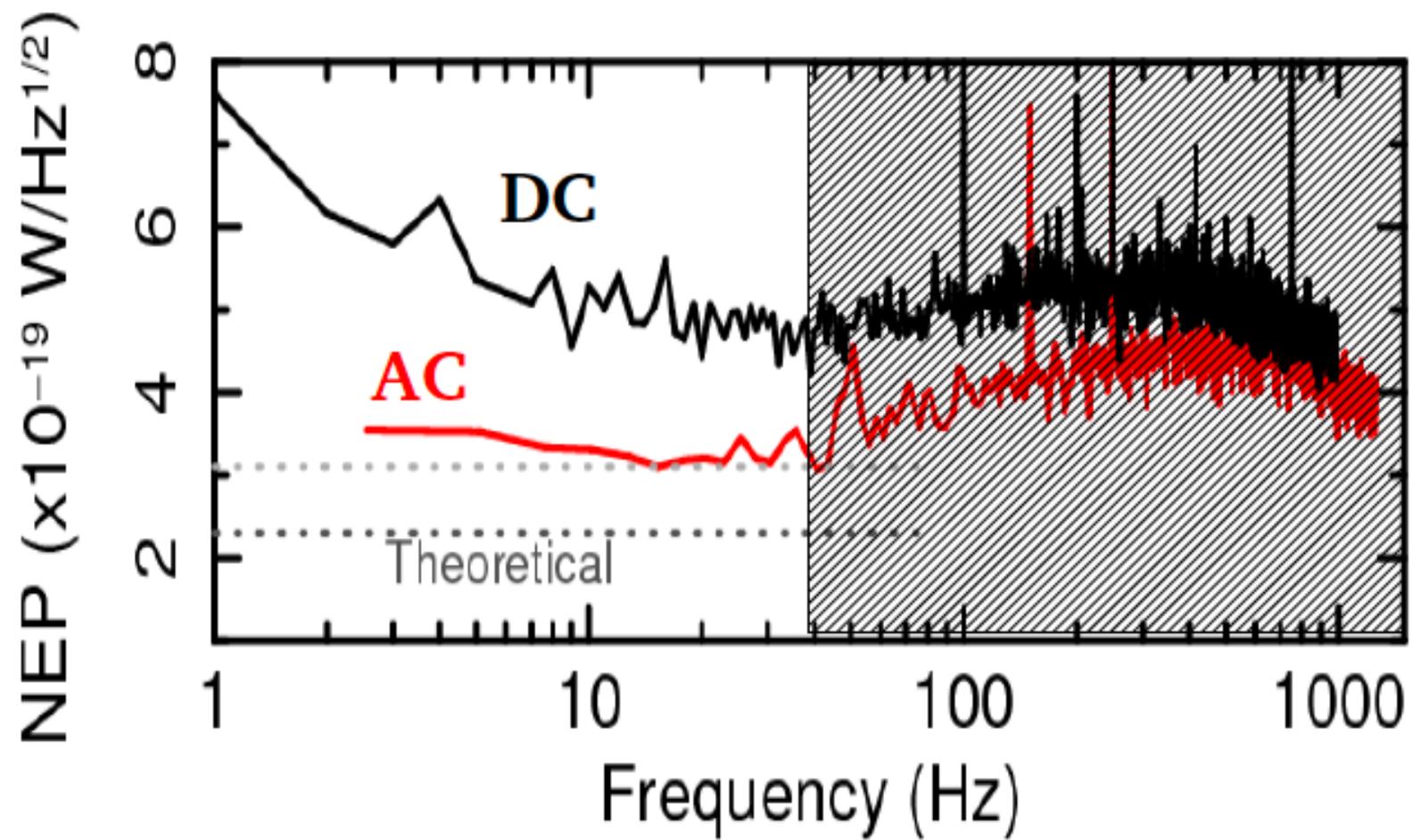
TES: Ti/Au $50 \times 50 \mu\text{m}^2$
thickness SiN
island/legs=500nm
 $G = 0.27 \text{ pW/K @ } 100\text{mK}$
 $T_c = 80.0 \text{ mK}$

Gao, et al

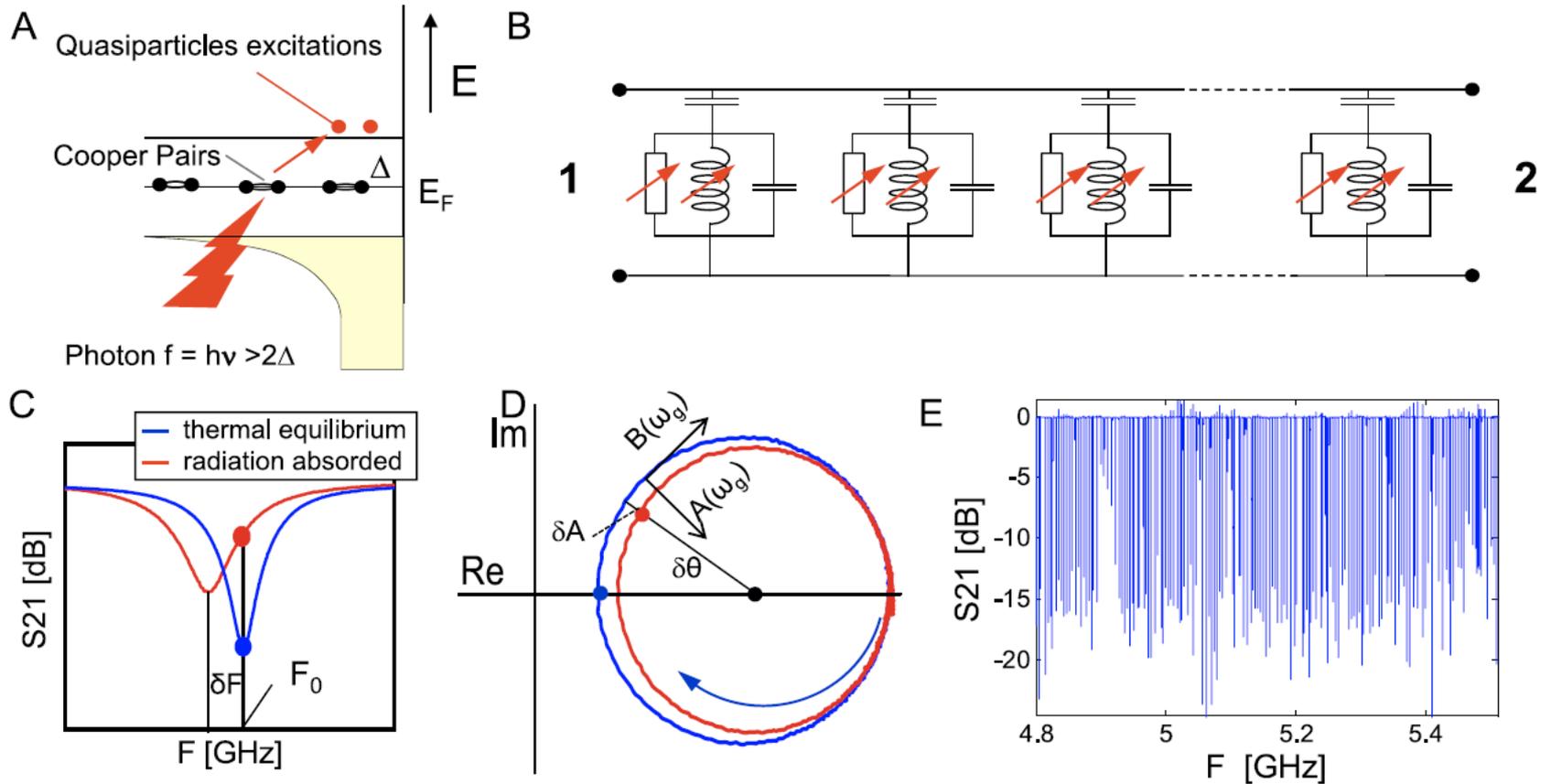




Measured sensitivity (50mK)

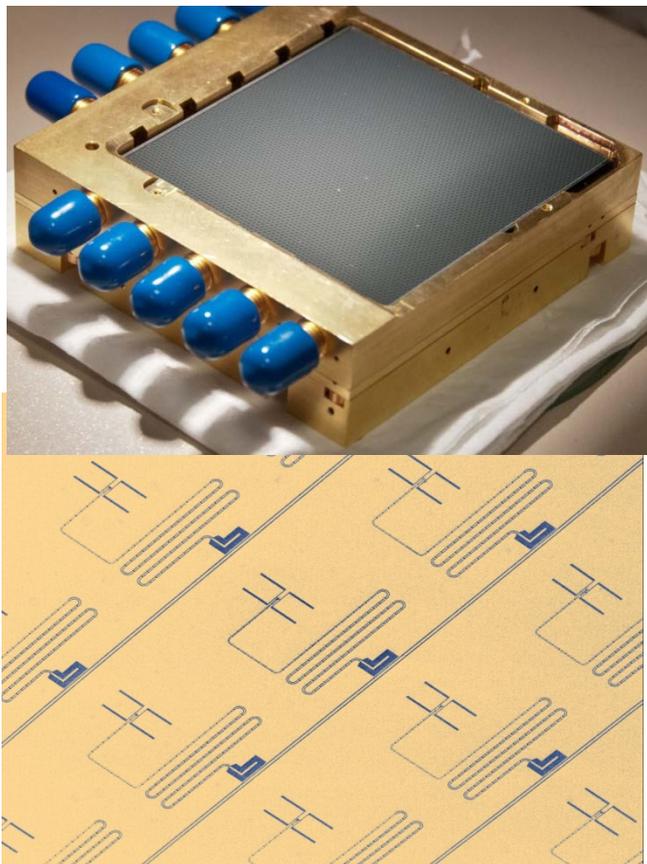


Microwave Kinetic Inductance detectors



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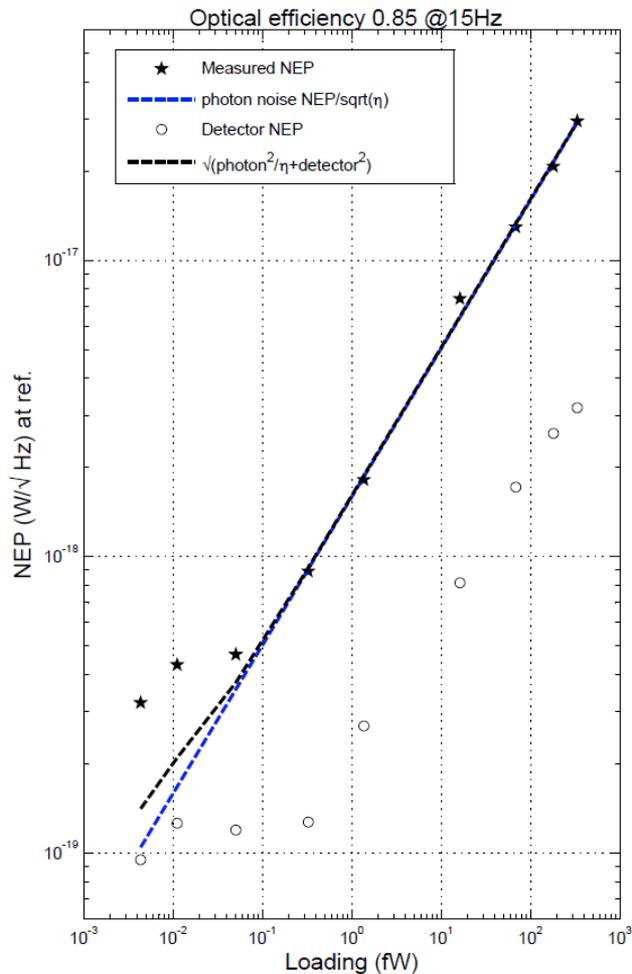
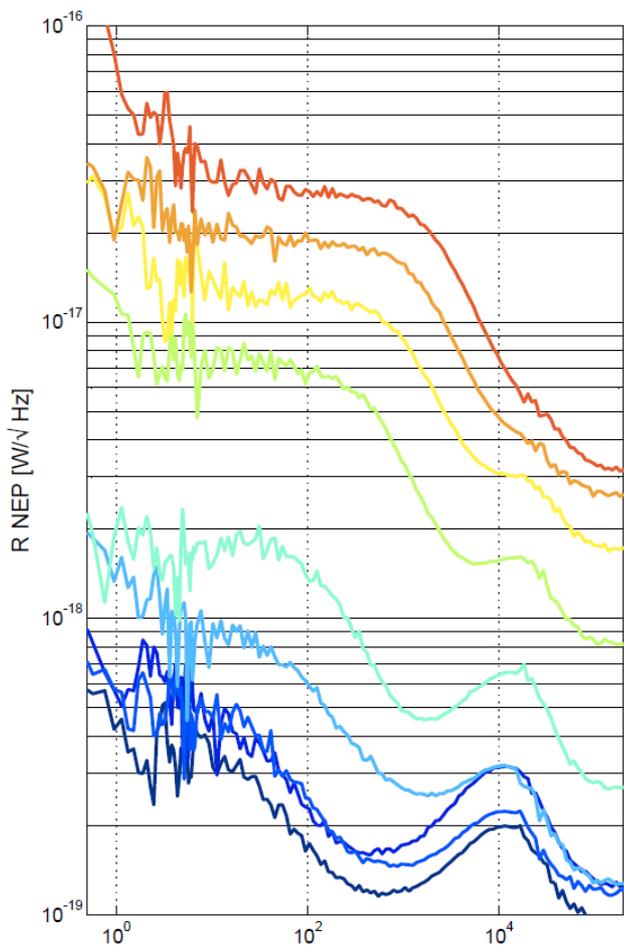
Ground based camera



25000 pixels, NEP $1E-18$ demonstrated $<1THz$



NEP $1E-19 > 1THz$

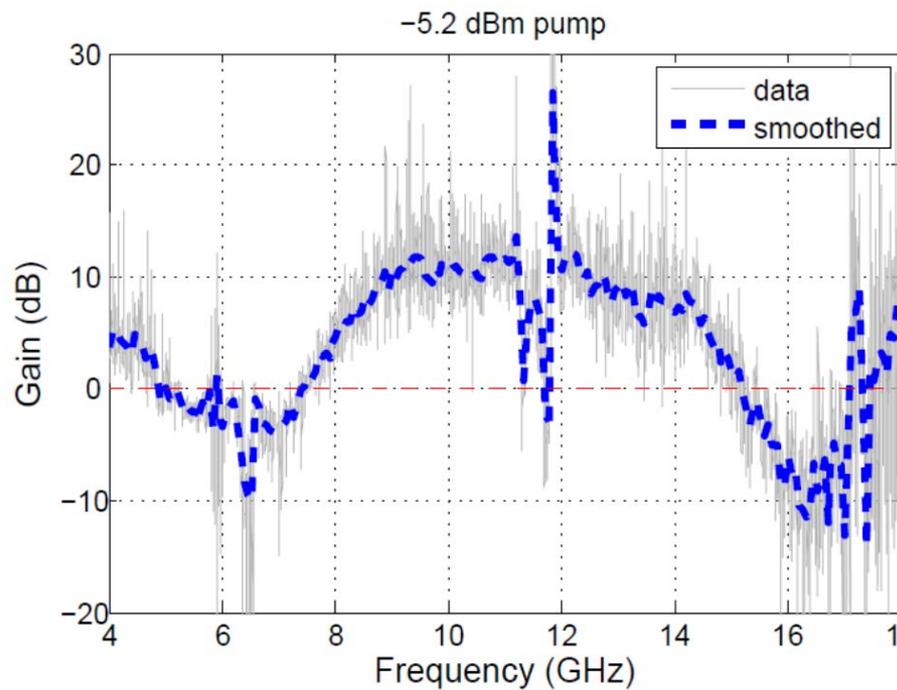
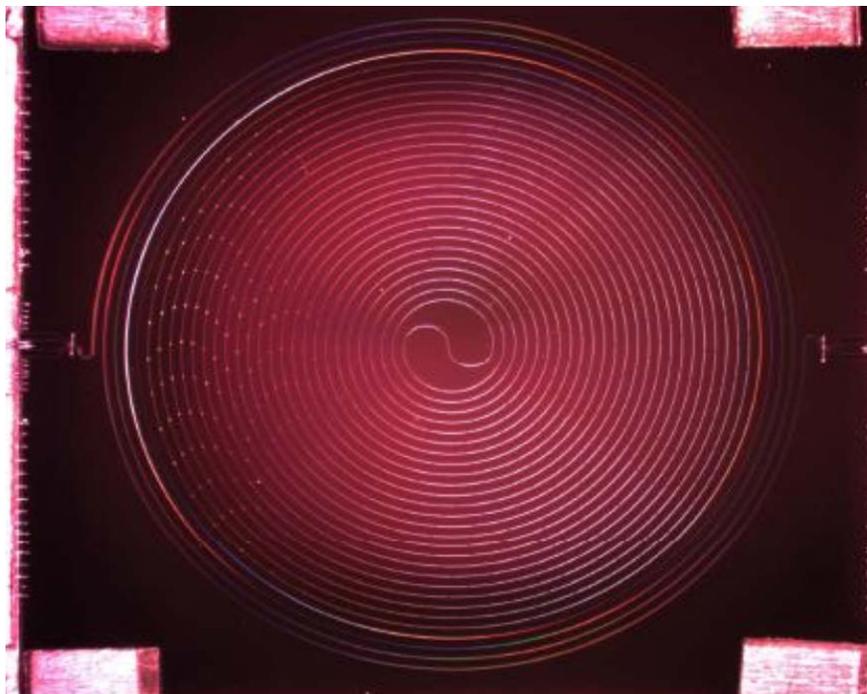


Baselmans et al





Parametric Amplifier Development



P. Day, J Zamuidzinas et al

